Inter-dependencies between emissions of CO$_2$, NO$_x$ & Noise from aviation

Policy Discussion Paper
September 2010

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Executive Summary

The aviation industry takes extremely seriously its responsibility to reduce its environmental impact, and has worked tirelessly in this regard for many decades. Over the last half-century, fuel-burn per passenger-kilometre has been reduced by some 70 per cent against a backdrop of progressively tightening noise and NO\textsubscript{X} regulations.

However, in the UK as elsewhere, the local environment agenda for aviation is driven largely by noise and occasionally by local air quality impacts, whereas the national and international agenda is primarily focussed on climate change. Addressing these often-competing demands is a constant challenge – achieving an improvement in one area may come at the expense of another. Understanding the implications involved in these inter-dependencies is crucial in the decision-making process. Clear and consistent advice from the regulators as to what should be the focus is essential in addressing the corresponding trade-offs.

This paper explores the nature of inter-dependencies between aviation’s emissions of CO\textsubscript{2}, NO\textsubscript{X} and noise\textsuperscript{1}, focussing on technological, operational and regulatory issues. Trade-offs\textsuperscript{2} against capacity are also discussed. We present a largely qualitative view of the inter-dependencies described, as their quantitative character is dependent on specifics at a level of detail beyond the paper’s scope.

We identify situations in which measures to reduce noise can increase fuel-burn, and circumstances in which measures to reduce fuel-burn may present challenges to meeting noise and NO\textsubscript{X} regulations. It is the purpose of this paper to raise awareness of these trade-offs and their implications for regulatory decisions, rather than to propose solutions to the difficulties they may present. However, we also identify measures that - for certain phases of the flight-cycle - are able to reduce CO\textsubscript{2}, NO\textsubscript{X} and/or noise with little or no trade-off. We report briefly on progress towards their development and/or deployment.

We emphasise that in this highly regulated industry safety is paramount and will never be compromised for technical or operational advantage.

Key messages arising from this document are as follows:

1. All parts of the aviation industry take their environmental responsibilities seriously, as demonstrated by significant improvements in environmental performance over many years.

2. Inter-dependencies between noise, NO\textsubscript{X} and CO\textsubscript{2} emissions are complex and require careful evaluation prior to regulatory, operational or design decisions. As regulations become more stringent, the relevant trade-offs become more difficult to address.

\textsuperscript{1} The impact of these emissions on climate or on local air quality is outside the scope of this paper.

\textsuperscript{2} In this paper we use the term “inter-dependency” to refer to a situation where a change in quantity A results in a change to quantity B. The more specific term “trade-off” is used to describe an inter-dependency where the changes in quantities A and B are in mutually opposing directions with respect to desirability i.e. if A is made “better”, B is likely to become “worse”.

2
3. Many of the trade-offs involve decisions by more than one section of the industry, making it essential that the various stakeholders work closely together to identify and promote good practice.

4. Local noise regulations can in some cases result in increased fuel-burn and CO₂ emissions arising from operational choices necessary to achieve compliance. Limitations in airspace or airport capacity can increase noise and emissions through holding, or through non-optimal cruising speeds or flight altitude profiles.

5. Some operational techniques have the potential to reduce noise, NOₓ and/or CO₂ emissions with no trade-offs. These include Continuous Decent Approach (CDA), Reduced-Engine Taxiing, and the use of Fixed Electrical Ground Power (FEGP) and/or Pre-Conditioned Air (PCA).

6. Engine fuel-efficiency can be improved by designing the core of the engine to run at increased temperatures and pressures. However, this presents challenges for managing NOₓ emissions. Successive generations of combustor designs have incorporated technologies to limit the peak gas temperatures and the duration of exposure, with the aim of limiting NOₓ emissions.

7. Compared with conventional turbofan configurations, open-rotor engine architectures offer significant potential for reduced fuel-burn and CO₂ emissions. Although open-rotor engines are likely to be quieter than today’s turbofans, a trade-off exists between their fuel-burn advantages and the noise-reduction potential of future turbofan designs.

8. More generally, noise regulations strongly influence the design of engines, effectively narrowing the design space and impacting on fuel-burn. For example, engine design parameters for the Airbus A380 were influenced by the requirement to meet QC/2 departure noise levels at London airports, to the slight detriment of fuel-burn and thus CO₂ emissions.

9. Noise regulations have not up to this point directly influenced airframe design to such an extent as to increase fuel-burn. However, looking forwards, more stringent noise regulations could lead to fuel-burn penalties arising from the need to incorporate additional airframe noise-reduction design features which result in increased weight and/or drag.

10. In general, it is not possible to express an inter-dependency in terms of a universally applicable metric, since its strength and character depend on the particular design point or operating point at which it is evaluated.

11. Regular dialogue between regulators and industry stakeholders will be essential in ensuring that the complexities of the topic, and the delicate balances required, are adequately accounted for in the decision-making process.
1 Introduction

The aviation industry has striven for many decades to reduce emissions of carbon dioxide (CO₂), noise, oxides of nitrogen (NOₓ) and other substances. Whilst NOₓ and noise³ are subject to regulatory limits, carbon dioxide emissions are directly related to fuel-burn and so a very strong commercial incentive has continued to drive greater efficiency through technological and operational improvements. Over the last half-century, fuel-burn per passenger-kilometre has been reduced by some 70 percent against a backdrop of progressively tightening noise and NOₓ regulations.

The industry remains resolute in its drive to reduce emissions even further, as evidenced by significant investment in ongoing research and development [SA, 2009], [SBAC, 2008], and by the European aviation industry’s commitment to achieving challenging targets set by ACARE. By 2020 we aim to develop technologies and operational practices which, when deployed in the subsequent years, will allow us to reduce fuel-burn per passenger-kilometre by 50 percent, to reduce NOₓ emissions by 80 per cent, and to reduce noise by half. These achievements will be benchmarked against equivalent new aircraft in 2000.

³ and also smoke, unburned hydrocarbons and carbon monoxide
Achievement of any one of these three targets would be challenging, but to achieve all three simultaneously will require considerable ingenuity and a clear understanding of the interdependencies between these three key drivers. This paper investigates these interdependencies, and catalogues the impact on each driver of potential technologies or operational practices.

We emphasise that safety is the number one priority of the aviation industry and will never be compromised for technical or operational advantage.

Emissions from aviation also include water-vapour, particulates, carbon monoxide, unburned hydro-carbons, soot and oxides of sulphur (SO\textsubscript{x}). The climate impacts of NO\textsubscript{x}, SO\textsubscript{x}, particulates, soot and water-vapour emissions are discussed in a separate paper [SA, 2008]. The current paper focuses on the inter-dependencies between emissions of CO\textsubscript{2}, NO\textsubscript{x} and noise.

In this paper, we start by exploring technological inter-dependencies, identifying for airframe and engine technologies the nature of the interactions between NO\textsubscript{x}, noise and CO\textsubscript{2} emissions. We then focus on operational aspects and the various techniques that may be employed to reduce noise or emissions, identifying the relevant inter-dependencies. We close with a discussion, in which we highlight “win-win” technologies or operational practices and assess progress towards their development and/or deployment.

As an aid to the reader, a glossary of terms is provided in Appendix D. Terms explained therein appear in italics in the main text.

2 Technological Inter-Dependencies

In this section we explore inter-dependencies related to engine and airframe design choices. In general terms, reductions in weight or drag reduce the requirement for thrust and are therefore beneficial with respect to noise, CO\textsubscript{2} and NO\textsubscript{x} emissions. However, significant trade-offs between CO\textsubscript{2} and NO\textsubscript{x}, and between CO\textsubscript{2} and noise, are evident in engine design choices, whilst future options for reducing airframe noise may carry fuel-burn penalties, as described below.

2.1 Engine

Although lower weight and reduced drag are generally advantageous for fuel-burn, NO\textsubscript{x} emissions and noise, many technological options for improving engine fuel-efficiency have different impacts on each of these three main drivers.

The efficiency of a jet engine can be characterised by two main factors. Firstly, the engine’s thermal efficiency describes the effectiveness with which the available chemical energy in the fuel is turned into mechanical energy. Secondly, the propulsive efficiency indicates how well the mechanical energy is turned into thrust. Higher values for both of these factors are desirable in the drive to reduce fuel-burn and CO\textsubscript{2} emissions.

Thermal efficiency is influenced primarily by the increase in pressure experienced by the air as it travels through the compressor, and by the temperature of the gas stream as it enters the turbine. A higher overall pressure ratio (OPR) and a higher turbine entry temperature (TET) both drive greater thermal efficiency. However, assuming a constant level of combustor technology, they also involve higher peak temperatures and chemical reaction rates during combustion, enhancing NO\textsubscript{x} formation.
Achieving higher thermal efficiencies without increasing total NO\textsubscript{X} emissions represents a significant challenge to engine designers, who must also ensure combustor operability under a wide range of conditions. Successive generations of combustor designs have incorporated technologies to limit the peak gas temperatures and the duration of exposure, against a backdrop of steadily rising overall pressure ratios. Options for the future include the use of intercooling - in which air is cooled before entering the final compressor stages, resulting in a lower combustor inlet temperature – and lean-burn combustion, in which the air-fuel ratio is higher than in more conventional arrangements. However, in comparison with more conventional combustor designs, the lean-burn concept is characterised by slightly lower efficiency and slightly higher weight, leading to slightly increased CO\textsubscript{2} emissions.

Propulsive efficiency is influenced primarily by the engine’s bypass ratio. Air entering the engine is split between the hot core and the cool bypass duct. A higher bypass ratio indicates that a smaller proportion of the total air intake is used within the core of the engine, and results in higher propulsive efficiency. However, for a given thrust rating, a higher bypass ratio typically requires a larger fan and its associated low-pressure turbine system, as well as a larger nacelle. Collectively these introduce extra weight and drag, potentially offsetting some or even all of the gains in propulsive efficiency. This trade-off between propulsive efficiency and weight/drag gives rise to an optimum bypass ratio - from the point of view of fuel-burn - for a given application.

The open-rotor architecture is an alternative engine configuration in which the bypass duct is omitted altogether, allowing for significantly higher bypass ratios without incurring the weight and drag penalty normally associated with a large nacelle. As well as its potential for improved fuel-burn, the open-rotor architecture offers benefits for lower NO\textsubscript{X} emissions, as it allows for a large increase in thrust without a corresponding increase in combustor temperatures.

However, the open-rotor configuration raises some challenges for the designer with respect to noise. For instance, the absence of fan-intake and bypass duct reduces opportunities for sound absorption, while the lack of intake flow-conditioning increases the likelihood that the leading rotor will experience non-uniform inflow arising from wing or airframe wake, leading to additional noise challenges. The interaction of the first rotor’s outflow with the second rotor presents a further potential source of noise.

Significant research has determined that the worst
effects of these noise challenges can be overcome, and that future open-rotor engines are likely to be quieter than today’s turbofans. Nonetheless, a trade-off exists between the fuel-burn advantages of a future open-rotor engine configuration, and the noise-reduction potential of future turbofan designs.

More generally, noise regulations strongly influence the design of engines, effectively narrowing the design space and potentially impacting on fuel-burn. For example, noise constraints may drive the choice of fan diameter to a larger size than would be chosen if fuel-burn were the sole driver, with additional implications for airframe configuration, as discussed in section 2.2.2 below. The design of the low-pressure turbine is also strongly constrained by noise considerations.

2.2 Airframe

Reducing the environmental impacts of aircraft has long been a priority for the aviation industry. In general, reduced fuel-burn, NO\textsubscript{X} emissions and noise are made possible by lower thrust requirements, enabled by lower airframe weight and reduced drag. These are achieved through the use of improved aerodynamics, advanced lightweight materials and systems, and improved design integration.

2.2.1 Trade-offs

The initial airframe design is constrained not only by trade-offs between CO\textsubscript{2}, NO\textsubscript{X} and noise, but also by market requirements such as design-speed, design-range, size and capacity. The aircraft manufacturer’s own vision of societal, including environmental, expectations over the aircraft’s lifetime – up to 30 years from entry into service – must also be factored in.

Based upon these considerations, manufacturers perform an evaluation starting well before a programme is launched, with a feasibility phase identifying design requirements and their potential solutions. An understanding of the environmental inter-dependencies at the earliest opportunity during an aircraft’s development is imperative for the aircraft manufacturer, in order to avoid any unintended consequences at a later stage. Many different aspects need to be considered to select the optimal aircraft configuration, for example: environmental impacts, performance, operability, reliability, system/component integration, operations, economics, market acceptance etc. All these aspects are inter-related and trade-offs will occur working towards an adequate and complex balance between them, subject to the overarching requirement of safety.

When integrating technologies that deliver improved efficiency, the aircraft’s overall fuel-burn saving potential is not equal to the sum of the individual technologies, due to interactions between components. Consequently, the design of individual components or sub-systems must be guided by their net benefits after integration effects and trade-offs have been taken into account. For example, an innovative new system may reduce drag at the expense of increased weight or power requirements.

2.2.2 Engine Integration

Many trade-offs at the engine level also have an impact on the airframe. For example, achievement of QC/2 departure noise levels at London airports resulted in a stronger weighting of noise within the overall design requirements of the A380 aircraft, with consequences for engine design choices. The availability of novel technologies and noise prediction capabilities allowed the achievement of the required acoustic performance of the aircraft with a limited trade against the potential range performance, acceptable to the market.
2.2.3 Noise

Airframe design up to now has not been directly affected by noise regulations to the detriment of fuel-burn. In general, the airframe has been designed for low fuel-burn and then low noise features have been incorporated if they do not cause a significant impact on performance. However, as noise regulations become progressively tighter, future options for reducing airframe noise still further may involve increased weight and/or drag. For example:

- In comparison with leading-edge slats used on today’s aircraft, alternative high-lift devices with lower noise characteristics may be less efficient at generating lift, and may lead to a requirement for a larger wing, with increased weight and drag.
- Shielding landing gear to reduce airframe noise, or using novel low-noise drag devices to improve steep approach capability in an effort to reduce noise impact, could increase aircraft weight.
- The airframe could be used to provide some kind of shielding to the engine noise, which may penalise the fuel-burn performance of the aircraft due to increased weight, increased drag, or impact on airflow into the engine.
- If noise regulations lead to engines with a larger-than-optimum bypass ratio, the engines may no longer be easily installed on to the airframe in the optimum position (under the wing with the wing beneath the fuselage floor). This may cause small fuel-burn penalties by affecting the length and weight of the landing gear or by increasing installation drag due to mounting the engine closer to the wing. Larger penalties may be caused if the engines need to be installed elsewhere (for example, high wing or rear fuselage mounted).

2.2.4 Conclusion

Aircraft manufacturers have worked to ensure that each new generation of aircraft has a lower environmental impact than previous generations. Noise levels, local air quality emissions and fuel-burn have all improved over the past years, though not necessarily at the same rates, or over the same timescales due to the changing emphasis on the individual environmental impacts from regulators and other stakeholder groups.

3 Operational Inter-Dependencies

The manner in which aircraft are operated has a significant bearing on noise impact, fuel-burn and NO\textsubscript{X} emissions. Trade-offs between these three drivers are relevant to many aspects of operational decision-making. Regulatory constraints – often driven by the need to minimise noise-impact – can bring these trade-offs into sharp focus. Appendix A describes some concrete examples of trade-offs in which noise-driven regulation has the potential to increase emissions of CO\textsubscript{2} and/or NO\textsubscript{X}.

For some phases of flight, such as approach, appropriate operational choices can result in environmental and other benefits without any trade-offs. However, for other phases such as take-off there are significant trade-offs, which must be carefully considered before following a particular procedure or technique. Capacity issues must also be taken into account.

To help identify and promote good practice within the aviation community, “Codes of Practice” have been developed both for arriving aircraft and for departures.

- The Arrivals Code of Practice [DfT, 2006] focuses on two techniques - Continuous Descent Approach (CDA) and Low Power, Low Drag (LPLD) - as methods for
reducing noise on approach. The additional benefits of these techniques are that they also reduce carbon dioxide and NO\textsubscript{X} emissions. Indeed, CDA was originally developed as a fuel conservation technique. The code is aimed specifically at air traffic controllers, flight crew and airports, and brings together advice for promoting these techniques from the combined expertise of DfT, NATS, BAA, airlines and CAA.

- The Departures Code of Practice focuses on four areas: gate operations (the use of Fixed Electrical Ground Power (FEGP) and Pre-Conditioned Air (PCA)); taxi-out with less than all engines operating; Continuous Climb Departures (CCD) and Collaborative Decision Making (CDM), as a way to maximise all operational efficiencies. The final version of the Code is scheduled for completion during 2010. In the meantime, an interim version has been released in two parts covering respectively reduced-engine-taxiing [ADS, 2009a] and FEGP/PCA [ADS, 2009b].

3.1 Air Traffic Management

For air traffic management the principal environmental trade-off is between noise and CO\textsubscript{2} emissions. The existing UK aviation policy that discourages, where possible, the over-flight of towns and cities and sensitive areas such as National Parks and Areas of Outstanding Natural Beauty in order to reduce noise exposure often results in additional track miles being flown with a resulting increase in fuel consumption and CO\textsubscript{2} emissions. Regulatory guidance for Airspace Change Proposals identifies the need to mitigate both noise and CO\textsubscript{2} emissions but provides no indication of circumstances under which noise or emissions should become the primary consideration.

Further examples of environmental trade-offs in air traffic management are included in the relevant operational sections below.

3.2 Airport Management

3.2.1 Airport Capacity

Airport expansion and runway capacity are contentious issues that can provoke intense public reaction. However, inadequate runway capacity requires certain operating procedures to be adopted which result in less favourable environmental performance.

Ground holding and airborne holding are particular practices necessary to achieve the required throughput at major airports constrained by runway capacity. Maximising throughput requires skilful sequencing of aircraft to minimise the separation distances between each movement and therefore necessitates a ‘reservoir’ of aircraft - both on the ground and in the air - to ensure that ATC always has a suitable mix of aircraft to sequence for take-off or landing with minimum gaps between. Clearly any delay incurred involves additional noise, NO\textsubscript{X} and CO\textsubscript{2} emissions.

Contrary to popular perception, there is therefore an environmental argument for increased airport and runway capacity at capacity-constrained airports, to minimise holding requirements for a given level of aircraft throughput. In the absence of additional runway
capacity, options for minimising ground-based holding at a given level of capacity constraint are discussed in section 3.3.3 below.

3.3 Ground Operations

Ground operations at airports include a number of activities that offer opportunities for environmental efficiencies in terms of minimising CO\textsubscript{2}, NO\textsubscript{X} and noise emissions, as detailed in this section. We also explore the issue of capacity, in particular the impact on environmental performance of inadequate system capacity.

3.3.1 Aircraft Operations on Stand

Aircraft require electrical power whilst on stand to support avionic systems, lighting, cabin air-conditioning, galley chillers etc. This electrical power would typically be provided from the aircraft’s Auxiliary Power Unit (APU). However, where available, the provision of electrical power and/or pre-conditioned air to the aircraft from airport infrastructure provides the opportunity to reduce not only CO\textsubscript{2} emissions but also NO\textsubscript{X} and noise, with no significant trade-offs. SA is working with airport members to promote the further deployment of Fixed Electrical Ground Power (FEGP) and Pre-Conditioned Air (PCA) in place of APU usage where appropriate [SA, 2010].

3.3.2 Taxiing

Taxiing on fewer than all engines is employed by some operators of some aircraft types at UK airports and has been shown to offer significant fuel and CO\textsubscript{2} reductions. It is estimated that through use of this technique, ground-level aircraft fuel-burn can be reduced by 20-40\% and ground-level aircraft NO\textsubscript{X} emissions by 10-30\% [ADS, 2009a]. SA and others are working to realise savings through the development of an industry code of practice [ADS, 2009a]. However, in some cases taxiing on a reduced number of engines may not be appropriate for safety reasons, particularly for aircraft with large engines from which jet blast may be a safety issue, for instance where there is work in progress within the airport boundary.

Trials of electric tugs to facilitate “engineless” taxiing have been very promising from an operational perspective - demonstrating that the procedure itself could be made to work, potentially reducing ground-level noise, NO\textsubscript{X} and CO\textsubscript{2} emissions with no trade-offs. However, it also identified some practical issues with towing aircraft, notably the potential to damage aircraft nose-gear over time. Work is ongoing to examine what potential exists to adapt aircraft to make them more resilient to towing.

3.3.3 Ground Holding

It is usual at capacity-constrained airports for aircraft to hold on the ground prior to departure to allow the efficient management of runway capacity. Technological improvements in managing and sharing ATC information offer potential in the future to reduce holding times and move towards gate-to-gate management of the flight cycle. The European coordinated SESAR programme is actively pursuing this goal through its development of the concept of Collaborative Decision Making (CDM). Reducing the pre-departure holding time offers the opportunity to reduce CO\textsubscript{2}, NO\textsubscript{X} and noise emissions without any trade-offs.

Nonetheless, if delay is to be incurred it is clearly preferable, from the point of view of fuel-burn, to absorb delay through holding on the ground rather than in the air. Work is underway to begin operational trials between Heathrow, Manchester and Amsterdam airports with the aim of reducing airborne holding, which could have significant emissions benefits.
### 3.4 Departure

A number of techniques can be applied in an attempt to reduce the environmental impact of aircraft during departure. It should be noted that some of the impacts will be highly dependent on the specific aircraft performance and the airport and runway characteristics, and legal constraints may limit what is actually available for use. There are significant trade-off issues here between noise, NO\textsubscript{X} and CO\textsubscript{2}, primarily related to the thrust-settings that may be employed at various stages of the departure procedure in an effort to minimise noise. Furthermore, improvements in noise levels may not be universal – even though noise levels may for instance be reduced under the flight path, the overall noise footprint may increase. Alternatively some techniques may reduce noise near the airport at the expense of increased noise further from the airport, coupled with increased fuel-burn. Appendix B lists the potential measures and the consequences of each in terms of noise, NO\textsubscript{X} and CO\textsubscript{2} emissions.

### 3.5 En-Route

The two principle operational variables that can influence en-route fuel-burn are flight-level (i.e. altitude) and cruising speed. Direct trade-offs between CO\textsubscript{2} and NO\textsubscript{X} emissions arising from these operational choices during the en-route phase of flight are not present. However, there are trade-offs between the choice of flight cruising speeds - driven in part by fuel-burn and airspace management considerations - and slot-timing constraints at airports driven both by airport capacity and by the need to minimise the noise disturbance caused by aircraft departures and arrivals. There are also trade-offs between airspace capacity and the ability to follow optimal speed and/or altitude profiles to minimise fuel-burn. Safety-driven constraints also play a role here. These issues are discussed below.

#### 3.5.1 Flight Level vs Airspace Capacity

On-board flight management systems and ground based flight planning software are used to identify the optimum cruising flight level, which will depend amongst other things on payload and mission-range, and varies as an aircraft burns fuel and reduces weight during its flight. Following a trajectory that differs from the optimum will result in increased fuel-burn and CO\textsubscript{2} production\textsuperscript{4}. By way of example, the fuel-consumption of an A340-600 aircraft flying 4,000 feet below its optimum flight level is around 2\% higher than at the optimum flight-level [Airbus, 2004].

Due to air-navigation rules, the ideal cruise-climb trajectory must be approximated by a sequence of step-climbs between flight-levels. Delaying a step-climb beyond its optimum point in the flight results in increased CO\textsubscript{2} emissions. In addition, ATM capacity restrictions and the requirement to protect ATC sectors from over-delivery of traffic can sometimes result in uneconomic flight-levels being allocated to operators, particularly in cases where the optimal flight-level profile has not been requested in flight-plans.

#### 3.5.2 Cruise Speed vs Airport Capacity and Noise Disturbance

The second variable is cruise speed. Aircraft have an optimum cruise speed or Mach number above or below which fuel efficiency is reduced. However, in response to a trade-off between fuel-costs and time-related costs, operators tend to fly slightly faster than the fuel-optimised cruise speed. The exact penalty will vary by aircraft type and journey length, but broadly

\textsuperscript{4} This has implications for some proposed contrail-avoidance strategies involving lower cruising altitudes.
speaking a difference of 0.01Mach above or below the optimum cruise speed can result in a trip fuel-burn penalty of the order of 0.2 – 0.5%.

Airport landing slots are based heavily on the expected flight times, which in turn are dependent to a great extent on cruise speed. As a result, variations in cruise speed can have a significant impact on the whole slot management system, especially at capacity constrained airports such as Heathrow.

Many airports have night restrictions or curfews in place in order to reduce noise impacts on local communities. Under these restrictions, changes to aircraft cruise speeds can result in holding or diversions due to arriving before a curfew ends, or after a curfew has commenced. This can give rise to a trade-off between local noise issues and flight-cycle fuel-burn.

3.5.3 Safety Constraints

From an air traffic control perspective, flight levels and cruise speeds are also important issues in managing the interaction of different aircraft flows. In some cases, a requested flight level cannot be offered due to the need to ensure safe separation from other aircraft. Similarly an air traffic controller may be required to instruct non-optimal speed in order to maintain safe separation and effectively manage flows, particularly when streaming aircraft ready for descent. The environmental ‘trade-off’ here is therefore with airspace capacity, driven by safety-related constraints.

3.6 Approach

Techniques such as Continuous Descent Approach (CDA), steeper, curved or decelerating approaches, and Low-Power-Low-Drag (LPLD), have been devised and tested by the industry. In general, the trade-offs between noise, NO\textsubscript{X} and CO\textsubscript{2} are less manifest for arrivals than they are for departures. Indeed, measures such as CDA have the potential to reduce all three simultaneously\textsuperscript{5}. Appendix C provides more details.

It should be noted that in some areas airspace capacity constraints may place limits on the simultaneous deployment of greener departures techniques and greener arrivals techniques, resulting in the requirement for a balance between the two.

4 Discussion and Summary

The UK aviation industry continues its active pursuit of opportunities to reduce its environmental impact. This paper has described many of the techniques that may be employed in the drive to reduce NO\textsubscript{X}, noise and carbon dioxide emissions from aircraft. However, as we have seen, the impact of an individual technique is not always beneficial with respect to all three of these often-competing drivers. Consequently the need for clear priorities must be borne in mind when proposing guidance or regulations. For instance:

- Noise regulations can stipulate or incentivise operational practices - such as alternative thrust profiles during take-off and climb - which may result in higher NO\textsubscript{X} emissions and/or fuel-burn.

\textsuperscript{5} However, CDA differs from conventional arrivals only at altitudes above 1000ft aal, and changes in NO\textsubscript{X} emissions above 1000ft aal have minimal impact on local ground-level NO\textsubscript{X} concentrations [ICAO, 2008].
Some airport noise regulations present design constraints for aircraft engines in particular, driving design choices that are not necessarily optimal in terms of fuel-burn.

Avoiding the overflight of populated areas in order to reduce noise exposure may result in increased fuel-burn arising from an increase in the actual distance flown.

In addition to the trade-offs between the three main environmental drivers of NO\textsubscript{X}, noise and CO\textsubscript{2}, implications for cost and capacity must also be taken into account when identifying suitable mitigation techniques. For example, in a capacity-constrained system, holding forms an essential part of maximising throughput by ensuring the availability of an aircraft for each landing or takeoff slot, or en-route airway. Whilst reduced holding is desirable from an environmental point of view, its achievement depends on the availability of adequate capacity for the whole system.

However, a number of operational techniques have the potential to reduce one or more of NO\textsubscript{X}, noise or carbon dioxide without significant detriment to the remainder or to cost and capacity. These include reduced-engine taxiing, towed-taxiing, fixed electrical ground power, continuous descent approaches, steeper approaches and low-power-low-drag approaches. Progress on these items includes the following:

- Towed taxiing has been explored in trials at Gatwick Airport, demonstrating that the concept may be feasible in the future. Issues surrounding the potential for damage to aircraft nose-gear, increased structural weight and additional taxiway congestion are still being worked.

- Reduced engine taxiing, already in use by many UK operators, is estimated to have the potential to reduce ground-level aircraft fuel-burn by 20-40% and ground-level aircraft NO\textsubscript{X} emissions by 10-30% [ADS, 2009a]. SA and others are working to realise this saving [SA, 2010].

- CDA has been adopted more widely across the UK over the past few years, as its benefits have been recognised and the relevant technological and procedural issues have been addressed. Further information can be found in SA’s 2009 Progress Report [SA, 2009].

- The use of FEGP and PCA in place of aircraft-mounted APU\textsubscript{s} has the potential to save CO\textsubscript{2}, NO\textsubscript{X} and noise with no significant trade-offs. SA is working to promote this practice [SA, 2010].

Improved aircraft aerodynamics and the reduction of aircraft weight represent significant opportunities to reduce thrust requirements, leading to reduced noise, NO\textsubscript{X} and carbon-dioxide emissions. Small reductions in weight can also result from operational choices e.g. installation of lighter seats.

In the UK, the local environment agenda for aviation is largely driven by noise and occasionally by local air quality impacts, whereas the national and international agenda is primarily focused on climate change. This presents a challenge for the aviation industry in addressing the often-competing demands of each of these issues. Greater awareness among policy makers of the potential down-stream effects of certain measures designed for environmental protection offers a starting point to addressing these. Environmental trade-offs should be considered early in any policy-making process in setting medium to long-term objectives to avoid unintended consequences.

This paper has largely avoided attempting to quantify the inter-dependencies that it has explored, on the grounds that to do so would require a detailed exposition of the particular technology levels, design choices and operational choices to which a particular trade-off applied.
Put more simply, the nature and strength of an inter-dependency typically varies according to the baseline from which it is evaluated, and so there is no single number expressing for instance the fuel-efficiency penalty arising from a desire to improve noise by 1dB, say. We therefore encourage regulators and policymakers to engage with industry stakeholders where appropriate to characterise and explore specific trade-offs relevant to proposed legislation.

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APPENDIX A: Operational Trade-off Issues in Practice

To illustrate the impact of trade-offs and the extent to which they are relevant to day-to-day decision-making across the industry, below we highlight some practical instances in which noise-driven regulation has the potential to increase emissions of CO$_2$ and/or NO$_X$.

**London airports - 1,000ft rule**

UK regulations for London airports state that after take-off, aircraft must achieve a height no less than 1000 ft aal at 6.5 km from start of roll (SoR), as measured along the departure track [NAR-LHR, 2010], [NAR-LGW, 2004], [NAR-STN, 2007]. This is to ensure that aircraft pass over the relevant noise monitor at “cut-back” power, whilst complying with international safety regulations concerning the minimum allowed “cut-back” height.

However, in order to reach 1,000ft aal at 6.5 km from SoR, some four-engined aircraft have to increase power above that normally used for take-off, resulting in greater NO$_X$ emissions and higher noise levels close to the airport. Engine maintenance costs may also be increased.

**London airports - departure noise limits**

UK regulations for London airports stipulate noise limits at specific monitor locations. In order to meet them, some operators have specified the use of higher than normal, or full rated, take-off power for departures to ensure that the aircraft is at a height where power cut-back (from that used for take-off) can legally be performed before reaching the relevant noise monitor.

As a result, although noise levels recorded by the monitor may be reduced, those close to the airport can be increased (especially to the side of the airport), and the noise contour area may be larger, depending on the noise characteristics of the aircraft. Additionally, the use of the maximum power setting increases NO$_X$ emissions significantly.

**Zurich - departure procedure**

At Zurich airport, the departure procedure is a legal requirement, intended to reduce noise by maintaining take-off power and a high-drag configuration for longer than would be the case under normal airline operating practice. Specifically, take-off power is stipulated until 2900 ft aal, and take-off flap setting is stipulated until 4500 ft aal [AIP-ZRH, 2005].

However, keeping the aircraft in a high-drag configuration until 4500 ft aal (vs. the more commonly used 1500 ft aal) results in higher fuel-burn than would otherwise be the case.

**Curfews**

Some airports such as Sydney [SACA, 1995] and Orange County [OCAIR] impose night-time curfews to limit noise impact on local communities. However, aircraft scheduled to arrive at such airports shortly after the curfew’s end risk increased fuel-burn if they arrive early. In such cases, aircraft may be held (resulting in additional fuel-burn and noise over the holding area), or diverted to another airport.

**Noise Preferential Routes**

Noise Preferential Routes (NPRs) are in common use at many airports for managing the noise impacts from aircraft departures. They are designed to concentrate aircraft over less
populated areas, thereby reducing the noise impact on areas of higher population density. In the UK, Government policy is to concentrate the aircraft noise along these routes, but airports in other countries may choose to adopt a dispersion approach, which spreads the effects more evenly. In both instances, where routes have been designed with sharp turns at low altitudes, or where they involve flying additional track miles, the potential for higher fuel burn and emissions will be greater.

There is also a trade-off between areas affected by noise. A concentration approach may result in fewer people being affected by aircraft noise, but may lead to a loss of tranquillity in areas where populations are sparser, such as wilderness areas.

Similar issues for arrivals may arise from the choice of ILS intercept location, and the noise associated with any manoeuvres that may be necessary to join ILS.

Discussion

As the above examples show, trade-offs between noise and fuel-burn or NO\textsubscript{X} are real and require careful attention when setting regulations in order to avoid unintended outcomes. Section 3 above discusses operational trade-offs in detail.
## APPENDIX B: Operational Inter-Dependencies – Departure

<table>
<thead>
<tr>
<th>Technique</th>
<th>Noise Impact</th>
<th>CO₂ Impact</th>
<th>NOₓ (LAQ) Impact</th>
<th>Comments</th>
</tr>
</thead>
<tbody>
<tr>
<td><strong>Increasing take-off power</strong></td>
<td>Reduces under flight-path, but footprint area can be increased</td>
<td>Slightly altered</td>
<td>NOₓ increases with power</td>
<td>Note 2. Adverse impact on engine maintenance costs</td>
</tr>
<tr>
<td><strong>Reducing take-off flap setting</strong></td>
<td>Reduces noise if lift-to-drag ratio improved - dependent on aircraft &amp; runway characteristics</td>
<td>May be slightly reduced</td>
<td>Slightly changed, dependant on aircraft &amp; runway characteristics</td>
<td>Note 2. Possible implications for tail strike under certain conditions</td>
</tr>
<tr>
<td><strong>NADP e.g. reduce acceleration altitude</strong></td>
<td>Noise increased close to airport, reduced further out</td>
<td>Reduced</td>
<td>Note 3</td>
<td>Note 4. Actual differences depend upon the difference in selected acceleration altitude versus standard airline practice.</td>
</tr>
<tr>
<td><strong>Delayed flap retraction in the climb</strong></td>
<td>Noise reduced close to airport, slight increase further out</td>
<td>Increased</td>
<td>Note 3</td>
<td>Note 4.</td>
</tr>
<tr>
<td><strong>Increased cut-back altitude</strong></td>
<td>Noise increased at some parts close to airport, reduced further out</td>
<td>Slightly reduced or increased, depending on flap retraction schedule.</td>
<td>Note 3</td>
<td>Note 4.</td>
</tr>
<tr>
<td><strong>Reduce power, retract flaps, then accelerate</strong></td>
<td>Reduced noise under flight-path, after normal acceleration point.</td>
<td>Increased</td>
<td>Note 3</td>
<td>Note 4. Aircraft in high-drag configuration with low power set may concern regulators.</td>
</tr>
<tr>
<td><strong>Increase VR, V2 and climb speeds</strong></td>
<td>Noise slightly increased close to airport, reduced further out</td>
<td>Very slightly increased</td>
<td>Increases due to the lower maximum thrust reduction available for take-off</td>
<td>Not applicable to some aircraft types and some operators. Depends upon takeoff performance limitations</td>
</tr>
<tr>
<td><strong>Increasing climb power settings</strong></td>
<td>Noise increases after cutback closer to the airport, reduces further out</td>
<td>Slightly reduced</td>
<td>Note 3</td>
<td>Note 4. Adverse impact on engine maintenance costs</td>
</tr>
<tr>
<td><strong>Novel Power Management (Managed Noise)</strong></td>
<td>Reduced at specific points identified as sensitive for noise.</td>
<td>Dependant on procedure, aircraft and airport requirements.</td>
<td>Note 3</td>
<td>Note 4. Currently only feasible with next generation aircraft, e.g. A380, A350, B787</td>
</tr>
</tbody>
</table>

**Note 1:** Although fuel flow is greater at the higher power setting, the time at that setting will be shorter, resulting in slight differences in overall fuel-burn that can be either positive or negative and will not be the same for all aircraft.

**Note 2:** Legal constraint: Noise Abatement Departure Procedures (NADPs) are not allowed below 800 ft (PANS-OPS/EU-OPS).

**Note 3:** Changes in NOₓ emissions above 1000ft aal have negligible impact on local ground-level NOₓ concentrations [ICAO, 2008].

**Note 4:** Will have an impact on flight path and speeds, so will need to keep ATC advised, and may affect adherence to Noise Preferential Routes with low level turns.
## APPENDIX C: Operational Inter-Dependencies – Approach

<table>
<thead>
<tr>
<th>Technique</th>
<th>Noise Impact</th>
<th>CO₂ Impact</th>
<th>NOₓ (LAQ) Impact</th>
<th>Comments</th>
</tr>
</thead>
</table>
| Continuous Descent Approach (CDA) | Reduced      | Reduced    | Little or no difference | Note 5, Note 6  
Procedures need to be set up. Greatest benefit will occur when initiated at higher altitudes with more advanced navigation equipment. |
| Low Power/Low Drag (LPLD)         | Reduced closer to the runway threshold | Reduced | Slight reduction | Note 6, Note 7, Note 8  
ICAO-stabilised approach criteria may also act as a constraint. |
| Steep Approach                    | Reduced      | Reduced.   | Note 9.         | Note 7, Note 8  
Legal constraint: Steep approach cannot be implemented solely for noise abatement purposes. [ICAO].  
LVP considerations may also limit application. |
| Curved Approach                   | Reduced      | Dependent on difference in track miles. | No difference below 1,000 ft aal | Note 5, Note 7  
Procedures need to be set up, and more advanced navigation equipment will be required. |
| Displaced or Inset Threshold (Note 10) | Note 9      | No difference | Note 9.         | Note 6, Note 8 |

Note 5: Reductions arising from these techniques are achievable above the ILS capture altitude. Below ILS capture, there is no noise or emissions benefit relative to standard approach.

Note 6: Safety considerations might preclude reductions in flap setting if runway is short or wet/contaminated.

Note 7: May require specialist aircraft and/or ground equipment to be installed, as well as additional training for aircrews.

Note 8: May result in increased use of reverse thrust, potentially eroding some of the benefits of the technique.

Note 9: Slight reduction in area impact, since low-level noise/emissions take place closer to (or within) the airport boundary.

Note 10: Moving the threshold along the runway so that it is further within the airport boundary.
APPENDIX D: Glossary

aal - above aerodrome level – the height above the aerodrome runway datum.

ACARE - Advisory Council for Aeronautics Research in Europe

ATC – Air Traffic Control

ATM – Air Traffic Management

Area Navigation (RNAV) – a method of navigation permitting flight in any airspace without the need to fly directly over ground-based navigational aids

APU - Auxiliary Power Unit – a unit installed on the aircraft. Provides electrical and pneumatic power to the aircraft on the ground

BAA – the company operating Heathrow, Stansted and other airports

CAA - Civil Aviation Authority – the UK’s aviation regulator

CO₂ - Carbon Dioxide – a greenhouse gas emitted by the burning of fossil fuels, including jet fuel

CDM - Collaborative Decision Making – a process of sharing the right information at the right time in the right place so leading to reduced airport delays and enhanced punctuality

CCD - Continuous Climb Departure – the use of a departure climb profile without level flight segments, allowing reductions in noise and fuel-burn/CO₂ compared with “stepped” climb departures

CDA - Continuous Descent Approach – the use of a descent profile without the level flight segments of the more traditional “stepped” descent. Has the potential to reduce noise and fuel-burn where used

DfT - Department for Transport – the UK Government department with responsibility for air transport

EU-OPS – The EU regulations specifying minimum safety and related procedures for commercial passenger and cargo fixed-wing aviation, published as Council Regulation (EEC) No 3922/91

FEGP - Fixed Electrical Ground Power – electrical power supplied from airport infrastructure rather than from an aircraft’s APU

ICAO – International Civil Aviation Organisation - the body that sets international standards and recommended practices for civil aircraft operations

ILS - Instrument Landing System – the electronic system giving guidance to the aircraft operating crews for the vertical (glide-slope) and horizontal (localiser) approach path at an airport

Landing and Takeoff Cycle – a phrase used to encapsulate the following phases of a flight: approach (from 3000 ft) and landing, taxi to/from the airport terminal, takeoff, and climb-out to 3000ft

LP/LD - Low Power/Low Drag – refers to the configuration and thrust of the aircraft while descending, i.e. in a low drag configuration (reduced flap and landing gear lowered as late as feasible) requiring low thrust levels to keep the required descent angle

LVP - Low Visibility Procedures – procedures put in place where the aerodrome wishes to continue operating instrument approaches where poor visibility or low cloud conditions are present

Mach number (Mach, or M) – the ratio of the velocity of the aircraft to the local speed of sound
Nacelle – the structure around the engine that forms its visible outer surface when on the aircraft

NADP - Noise Abatement Departure Procedure – the procedure used during take-off and initial climb to reduce the noise impacts beneath the aircraft

NATS – provider of air navigation services in UK airspace and the eastern part of the North Atlantic

NPR - Noise Preferential Route – the departure route from an airport’s runway that allows for the minimum noise impact to local communities on the ground. Sometimes called Minimum Noise Routes

OPR - Overall Pressure Ratio – a measure of the extent to which air entering the engine is compressed before entering the combustor

NO\textsubscript{X} - Oxides of Nitrogen – gases produced when air is raised to very high temperatures. NO\textsubscript{X} emissions have consequences for local air quality, and are also believed to impact the atmospheric concentrations of methane and ozone (both of which are greenhouse gases)

PANS-OPS – Procedures for Air Navigation Services - Aircraft Operations - outlines the principles for airspace protection and procedure design to which all ICAO signatory states should adhere. Published as ICAO Doc 8168

PCA - Pre-Conditioned Air – a source of heated or cooled air for aircraft cabin air-conditioning supplied from airport infrastructure rather than from an aircraft’s APU

P-RNAV - Precision RNAV – an aircraft navigation standard that requires a track-keeping accuracy of ± 1 nautical mile for at least 95% of the flight time

PLTOW – Performance Limited Take-Off Weight, the aircraft’s take-off weight for a particular set of conditions that is limited by only scheduled performance limitations (i.e. not structural considerations)

SID – Standard Instrument Departure - the air traffic control coded departure procedure instructions to an aircraft’s operating crew that should allow the NPR to be followed

SES - Single European Sky – a European initiative putting forward a legislative approach to solving the ATM issues currently affecting air transport, as well as enabling ATM to cope with future demands

SESAR - Single European Sky ATM Research

Steep Approach – an approach to landing at an airport that requires a final descent angle above 4.5°, though the term is sometimes used to refer to an approach greater than the normal maximum of 3.25°

Threshold – the start of the portion of the runway that is available for landing aircraft

TET - Turbine Entry Temperature – the temperature of the gases exiting the engine's combustor and entering the turbine

VR – the speed at which an aircraft is rotated to lift off the runway, preserving all the necessary safety margins for take-off speed

V2 – the Take-off safety speed that is, the speed which the aircraft has to attain at the end of the take-off phase at the 35ft screen height